

Hydrogen Plasma Core Transport Systems with Dual Counter-Rotating Magnetic Lenses for Cislunar Infrastructure

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Abstract:

This paper presents a theoretical framework for hydrogen plasma core transport systems featuring dual counter-rotating magnetic lenses. The architecture integrates propulsion, power generation, magnetic confinement, and navigation into a unified mechanism. The plasma core emits at the hydrogen hyperfine transition frequency of 1420.405 MHz, providing an intrinsic frequency reference requiring no external oscillators. Three orbital stations positioned at low Earth orbit, the lunar surface near Shackleton Crater, and Sun-Earth Lagrange Point 2 establish a navigation network through phase measurement of the standing wave pattern. The theoretical basis draws from two-fluid plasma transport theory, centrifugal mirror confinement validated by the Maryland Centrifugal Experiment, and recent advances in high-temperature superconducting magnets demonstrated by the Wisconsin HTS Axisymmetric Mirror experiment. Lunar surface installations leverage the natural cryogenic environment of permanently shadowed craters for superconducting magnet operation without active cooling. This paper derives testable transport scaling laws that differ from standard single-fluid predictions and proposes validation pathways using existing experimental facilities.

Keywords: hydrogen plasma, counter-rotating lenses, magnetic confinement, 1420 MHz, cislunar navigation, lunar infrastructure, two-fluid transport, centrifugal mirror.

1. Introduction

Establishing sustainable cislunar infrastructure requires transport systems that can operate autonomously over long durations with minimal dependence on Earth-based resources. Current spacecraft designs separate propulsion, power generation, attitude control, and navigation into distinct subsystems, each adding mass, complexity, and potential failure modes. This paper proposes an integrated architecture where a single mechanism—dual counter-rotating magnetic lenses surrounding a hydrogen plasma core—provides all four functions simultaneously.

The theoretical foundation combines established plasma physics with recent experimental advances. Two-fluid transport theory, developed by Braginskii in 1965 [1], describes how ions and electrons move through magnetized plasmas with extreme anisotropy between parallel and perpendicular directions. Centrifugal mirror confinement, demonstrated at the Maryland Centrifugal Experiment from 1998-2010 [2], showed that supersonic plasma rotation can suppress turbulent transport and achieve classical confinement scaling. High-temperature superconducting magnets, recently demonstrated at record 17 Tesla fields in the Wisconsin HTS Axisymmetric Mirror [3], enable compact, high-field magnetic configurations without active cryogenic systems when operated in naturally cold environments.

The hydrogen hyperfine transition at 1420.405751768 MHz provides the most precisely defined natural frequency in atomic physics [4]. A hydrogen plasma naturally emits at this frequency as atoms transition between spin states. This emission serves as an intrinsic frequency reference, enabling navigation through phase measurement of signals from fixed stations without requiring onboard atomic clocks or communication with ground control.

The lunar surface, particularly the south polar region near Shackleton Crater, offers unique advantages for plasma-based energy infrastructure. Permanently shadowed craters maintain temperatures of 20-40 Kelvin [5], cold enough for high-temperature superconductors to operate with zero electrical resistance. Adjacent sunlit ridges receive solar illumination up to 89% of the year [6]. Water ice deposits confirmed by the LCROSS mission at concentrations of 5.6 ± 2.9 weight percent [7] provide feedstock for hydrogen fuel production through electrolysis.

2. Theoretical Framework

2.1 Two-Fluid Transport in Magnetized Hydrogen Plasma

Plasma transport in magnetic fields exhibits extreme anisotropy. Charged particles spiral freely along magnetic field lines but cross them only through collisions or turbulent fluctuations. The Braginskii two-fluid equations describe this behavior by treating ions and electrons as separate fluids coupled through electromagnetic fields and collisions [1].

For a partially ionized hydrogen plasma, the ion momentum equation includes friction with neutral atoms:

$$m_i n_i (\partial V_i / \partial t + V_i \cdot \nabla V_i) = -\nabla p + (1/c) j \times B - \alpha (V_i - V_n)$$

where m_i is ion mass, n_i is ion density, V_i is ion velocity, p is pressure, j is current density, B is magnetic field, and α is the ion-neutral friction coefficient. The friction term couples the ionized plasma to neutral hydrogen atoms, which do not respond to magnetic fields.

Transport coefficients differ by many orders of magnitude between directions parallel and perpendicular to the magnetic field. For electrons, the parallel thermal conductivity scales as [1]:

$$\kappa_{\parallel} = 3.2 n \tau_e T_e / m_e$$

while perpendicular conductivity drops to:

$$\kappa_{\perp} = 4.7 n T_e / (m_e \Omega_e^2 \tau_e)$$

where τ_e is the electron collision time and Ω_e is the electron cyclotron frequency. The ratio $\kappa_{\parallel}/\kappa_{\perp}$ reaches 10^9 to 10^{12} in strongly magnetized plasmas [8], meaning heat flows almost exclusively along field lines.

Perpendicular particle diffusion follows two distinct scaling laws depending on transport mechanism. Classical diffusion, caused by binary collisions, scales inversely with the square of magnetic field strength:

$$D_{\text{classical}} \propto 1/B^2$$

Anomalous or Bohm diffusion, driven by turbulent fluctuations, scales only as $1/B$ [9]:

$$D_{\text{Bohm}} = (1/16)(k_B T_e)/(eB)$$

This scaling difference provides a critical experimental test. Achieving $1/B^2$ scaling indicates suppression of turbulence and classical confinement; $1/B$ scaling indicates turbulence-dominated transport.

2.2 Ambipolar Diffusion and Self-Generated Electric Fields

Electrons move faster than ions due to their smaller mass. In a plasma without external circuits, electrons cannot escape faster than ions without violating charge neutrality. The plasma generates an internal electric field that slows electrons and accelerates ions until both species diffuse at the same rate. This coupled motion is called ambipolar diffusion [10].

The ambipolar diffusion coefficient in unmagnetized plasma is:

$$D_a = D_i (1 + T_e/T_i)$$

where D_i is the ion diffusion coefficient and T_e/T_i is the electron-to-ion temperature ratio. When electrons are much hotter than ions (typical in laboratory plasmas where T_e/T_i can reach 50-100), ambipolar diffusion approaches twice the ion diffusion rate.

The self-generated ambipolar electric field is:

$$E_a = (k_B T_e/e)(\nabla n/n)$$

This field points down the density gradient, confining electrons electrostatically. In rotating plasma systems, this electrostatic potential combines with centrifugal effects to create an effective confinement barrier that traps both species.

2.3 Centrifugal Mirror Confinement

Magnetic mirror machines confine plasma between regions of high magnetic field. Particles with sufficient velocity parallel to the field escape through the mirror throats. The loss cone—the range of velocity directions that lead to escape—limits confinement time.

Plasma rotation adds centrifugal confinement. A plasma rotating at angular velocity ω experiences an effective outward gravity:

$$g_{\text{eff}} = \omega^2 r$$

where r is the radial distance from the rotation axis. If the magnetic field lines curve inward at the mirror throats, the centrifugal force pushes plasma away from the loss regions and toward the midplane.

The effective potential well depth scales with the square of the sonic Mach number [2]:

$$\Delta\Phi \sim M^2 T_e$$

At Mach 3, the potential well depth equals 9 times the electron temperature, providing exponential suppression of parallel losses. The Maryland Centrifugal Experiment demonstrated this physics with rotation velocities of 100-200 km/s, Mach numbers up to 2.5, and density drops of $10\times$ from midplane to mirror throats following Boltzmann statistics [2].

Critically, MCX achieved velocity shear 9 times above the threshold for suppressing interchange instabilities [2]. This shear decorrelates turbulent eddies, enabling classical rather than anomalous transport. The measured perpendicular diffusion followed $1/B^2$ scaling, confirming turbulence suppression.

2.4 Testable Prediction: Transport Scaling in Counter-Rotating Geometry

Counter-rotating plasma layers present a configuration not previously explored experimentally. At the interface between layers rotating in opposite directions, velocity shear doubles compared to single-rotation systems. Standard two-fluid theory predicts this enhanced shear will further suppress turbulent transport, reducing perpendicular diffusion below single-rotation values.

However, the shear layer also introduces Kelvin-Helmholtz instability risk. The stability criterion requires:

$$|\Delta V| < V_A$$

where ΔV is the velocity difference across the interface and V_A is the Alfvén velocity. For hydrogen plasma in a 1 Tesla field at density 10^{19} m^{-3} , $V_A \approx 220 \text{ km/s}$. Counter-rotation at Mach 2 in each direction (total shear of Mach 4) at ion temperature 100 eV gives $\Delta V \approx 160 \text{ km/s}$, below the Alfvén limit.

The testable prediction is that counter-rotating systems will exhibit perpendicular diffusion coefficients $D_\perp$ scaling as $1/B^2$ with magnitude reduced by a factor of 2-4 compared to single-rotation systems at the same Mach number, provided $|\Delta V| < V_A$ is satisfied. This can be tested in existing centrifugal mirror facilities by modifying electrode configurations to drive counter-rotation.

3. System Architecture

3.1 Transport Vehicle Configuration

The transport vehicle consists of three concentric systems arranged along a central axis. At the center, a hydrogen plasma core operates at temperatures of 1-10 eV with density around 10^{19} m^{-3} . The plasma continuously emits at 1420.405 MHz as hydrogen atoms undergo hyperfine transitions between spin states. Surrounding the plasma core, two magnetic lens assemblies rotate in opposite directions. Each assembly consists of superconducting coils embedded in rotating rings. The inner lens rotates clockwise when viewed from the sunward direction; the outer lens rotates counter-clockwise at the same angular velocity. The gap between the two assemblies is maintained by magnetic repulsion.

The counter-rotating lenses serve four functions simultaneously:

- **Solar concentration:** The lens assemblies focus incoming sunlight onto the plasma core, providing thermal energy input.
- **Magnetic confinement:** The rotating superconducting coils generate magnetic fields that confine the plasma through combined magnetic mirror and centrifugal effects.
- **Gyroscopic stability:** The angular momentum of the rotating assemblies resists orientation changes. Counter-rotation allows net angular momentum to be set to any value, including zero.
- **Dynamic focus control:** Differential adjustment of rotation speeds shifts the focal point, enabling modulation of solar concentration without mechanical actuators.

When both lenses rotate at matched speeds, their focal effects combine to produce a stable focal point. Adjusting one lens to rotate faster or slower than the other shifts the focus. This differential control enables continuous adjustment of energy input to the plasma core.

3.2 Navigation Network

Three fixed stations establish a navigation grid across cislunar space. Station One operates in low Earth orbit at approximately 400 km altitude. Station Two sits on the lunar surface near Shackleton Crater. Station Three orbits Sun-Earth Lagrange Point 2, approximately 1.5 million km from Earth.

Each station contains a hydrogen plasma core broadcasting at 1420.405 MHz. Because all stations emit at the same frequency, their signals interfere, creating a standing wave pattern throughout the navigation volume. The wavelength at 1420.405 MHz is 21.106 cm, giving the pattern structure at centimeter scales.

A transport vehicle determines position by measuring the phase of received signals from each station. Phase depends on distance: every 21.1 cm of path length adds 360 degrees of phase. Measuring phase from all three stations constrains position to a unique point.

The asymmetric geometry of the three stations—380,000 km (Earth to Moon), 1,400,000 km (Moon to L2), and 1,500,000 km (Earth to L2)—ensures that every location in cislunar space has a unique phase signature. Navigation is autonomous: the vehicle reads the ambient field pattern without requiring communication with ground control.

4. Lunar Surface Infrastructure

4.1 Optimal Surface Positioning

The lunar south pole offers the optimal location for plasma-based energy infrastructure. The connecting ridge between Shackleton and de Gerlache craters provides the best combination of solar illumination and proximity to water ice resources [6].

The optimal installation point lies at coordinates 89.128°S, 136.14°W at 1,947 m elevation, receiving solar illumination 85.7% of the year. Combining two stations 8 km apart increases combined illumination to 94% annually. Maximum eclipse duration for all Shackleton rim locations is 43 hours [6].

Permanently shadowed regions within meters to hundreds of meters of the illuminated ridge maintain temperatures of 20-40 K year-round [5]. These temperatures lie well below the critical temperature of high-temperature superconductors such as YBCO ($T_c = 93$ K) and REBCO ($T_c = 92$ K), enabling superconducting magnet operation without active cryogenic systems.

4.2 Plasma Ignition and Maintenance

Microwave electron cyclotron resonance (ECR) heating provides the optimal plasma ignition method for lunar vacuum conditions. At the ECR condition where microwave frequency matches electron cyclotron frequency (875 Gauss at 2.45 GHz), microwaves transfer energy efficiently to electrons [11]. Modern solid-state microwave sources achieve efficiencies exceeding 90%.

ECR heating operates effectively at pressures as low as 10^{-5} torr—well above the lunar ambient pressure of approximately 3×10^{-12} torr—requiring only modest hydrogen injection to sustain the plasma. The electrode-free design eliminates erosion concerns that limit arc discharge systems.

For hydrogen-oxygen combustion in controlled energy release, platinum group catalysts enable ignition at temperatures as low as 33-37°C [12]. Iridium catalysts reduce ignition delay by 50% compared to uncatalyzed systems. Laser ignition systems, demonstrated by Russian tests in 2018 [13], enable thousands of restart cycles without electrode wear.

4.3 Energy Storage for Lunar Night

The 14-Earth-day lunar night requires energy storage systems capable of sustaining operations for approximately 354 hours without solar input. NASA's Kilopower fission reactor provides the most viable continuous power solution, delivering 1-10 kW electrical output from a 134 kg unit containing 28 kg of uranium-235 fuel [14].

For supplemental storage, solid-state lithium batteries have reported laboratory energy densities as high as 824 Wh/kg in preliminary industry testing [23], though this figure comes from an industry announcement rather than a peer-reviewed measurement and should be treated as provisional. Imperia Batteries, under NASA contract, is developing a ceramic separator system specifically rated for the full lunar temperature range of -230°C to +120°C [16]—the only chemical battery technology currently claiming this specification.

4.4 L2 Orbital Collector

Sun-Earth Lagrange Point 2 receives continuous solar illumination at the full 1361 W/m² intensity, unattenuated by atmosphere. JWST demonstrates L2 operational viability with modest station-keeping requirements of approximately 2.5 m/s per year [17].

Direct energy transmission from L2 to the lunar surface faces fundamental physics constraints. At 1.5 million km distance, diffraction-limited laser beams expand to kilometer-scale spot sizes even with meter-class transmitter apertures. A relay architecture through Earth-Moon L1 (approximately 60,000 km from the Moon) or low lunar orbit (100 km altitude) improves transmission efficiency by orders of magnitude while L2 collectors feed the relay network.

5. Experimental Validation Pathways

5.1 Transport Scaling Validation

The primary theoretical prediction—reduced perpendicular diffusion in counter-rotating geometry—can be tested by modifying existing centrifugal mirror facilities. The Centrifugal Mirror Fusion Experiment (CMFX), which achieved first fusion neutrons in 2025 [18], provides an appropriate testbed.

Validation requires measuring perpendicular diffusion coefficient $D_{\perp}$ as a function of magnetic field B for both single-rotation and counter-rotation configurations at matched Mach numbers. Classical transport

predicts $D_{\perp} \propto 1/B^2$ while anomalous transport predicts $D_{\perp} \propto 1/B$. The proposed mechanism predicts counter-rotation will reduce $D_{\perp}$ magnitude by a factor of 2-4 at fixed Mach number while maintaining $1/B^2$ scaling.

5.2 High-Field Magnet Integration

The Wisconsin HTS Axisymmetric Mirror (WHAM) achieved first plasma in July 2024 with record 17 Tesla fields using REBCO high-temperature superconducting magnets [3]. WHAM parameters—adjustable central field (0.32-0.86 T), mirror ratios up to 50, 1 MW ECH heating—provide a platform for testing the proposed magnetic lens geometry.

Integration of rotating magnetic elements with WHAM-class high-field magnets would validate the combined centrifugal-mirror confinement concept at parameters relevant to the proposed transport system.

5.3 21-cm Emission Detection

The hydrogen hyperfine transition has an Einstein A coefficient of $2.88 \times 10^{-15} \text{ s}^{-1}$, corresponding to a spontaneous decay lifetime of 11 million years [4]. Direct detection of 21-cm emission from laboratory plasmas requires maser amplification through population inversion.

Hydrogen masers achieve fractional frequency stability of 10^{-15} at 1000 seconds averaging time [19]. Adapting maser cavity techniques to the transport vehicle plasma core would validate the frequency reference capability. The critical test is demonstrating phase coherence between separated plasma sources at the level required for navigation (approximately 1 degree phase uncertainty corresponds to 6 cm position uncertainty).

6. Conclusions

This paper presents a theoretical framework for hydrogen plasma core transport systems with dual counter-rotating magnetic lenses. The architecture integrates propulsion, power, confinement, and navigation into a single mechanism, reducing system mass and complexity compared to conventional designs with separate subsystems.

The theoretical basis draws from established plasma physics—Braginskii two-fluid transport, ambipolar diffusion, centrifugal mirror confinement—while proposing a specific testable prediction for counter-rotating geometry: perpendicular diffusion coefficients reduced by a factor of 2-4 compared to single-rotation systems at matched Mach numbers, with $1/B^2$ scaling maintained.

Lunar surface installations leverage natural environmental advantages: cryogenic temperatures in permanently shadowed craters for superconducting magnets, near-continuous solar illumination at polar ridges, and water ice deposits for hydrogen fuel production. The L2 station provides continuous solar collection with demonstrated operational viability.

Validation pathways exist through modification of existing facilities including CMFX for transport scaling measurements, WHAM for high-field magnet integration, and hydrogen maser techniques for frequency reference demonstration. The counter-rotating interface stability criterion $|\Delta V| < V_A$ provides a clear engineering constraint for system design.

REFERENCES:

1. Braginskii, S. I. (1965). Transport processes in a plasma. *Reviews of Plasma Physics*, 1, 205-311.
2. Ellis, R. F., Case, A., Elton, R., et al. (2005). Steady supersonically rotating plasmas in the Maryland Centrifugal Experiment. *Physics of Plasmas*, 12, 055704.
3. Endrizzi, D., Anderson, J., Brown, M., et al. (2023). Physics basis for the Wisconsin HTS Axisymmetric Mirror (WHAM). *Journal of Plasma Physics*, 89, 975890501.
4. Ewen, H. I., & Purcell, E. M. (1951). Observation of a line in the galactic radio spectrum: Radiation from galactic hydrogen at 1,420 Mc./sec. *Nature*, 168(4270), 356.
5. Paige, D. A., Siegler, M. A., Zhang, J. A., et al. (2010). Diviner Lunar Radiometer observations of cold traps in the Moon's south polar region. *Science*, 330(6003), 479-482.
6. Gläser, P., Oberst, J., Neumann, G. A., et al. (2018). Illumination conditions at the lunar poles: Implications for future exploration. *Planetary and Space Science*, 162, 170-178.
7. Colaprete, A., Schultz, P., Heldmann, J., et al. (2010). Detection of water in the LCROSS ejecta plume. *Science*, 330(6003), 463-468.
8. Fundamenski, W. (2010). *Power exhaust in fusion plasmas*. Cambridge University Press.
9. Bohm, D. (1949). *The characteristics of electrical discharges in magnetic fields*. McGraw-Hill.
10. Chen, F. F. (2016). *Introduction to Plasma Physics and Controlled Fusion* (3rd ed.). Springer.
11. Lieberman, M. A., & Lichtenberg, A. J. (2005). *Principles of Plasma Discharges and Materials Processing* (2nd ed.). Wiley-Interscience.
12. Ryu, J. H., Lee, K. Y., La, H. J., et al. (2016). Role of hydrogen and oxygen activation over Pt and Pd-doped composites for catalytic hydrogen combustion. *ACS Applied Materials & Interfaces*, 8(40), 27332-27339.
13. TASS. (2018). Russia tests laser ignition for oxygen-hydrogen rocket engine for first time. TASS Russian News Agency.
14. Gibson, M. A., Oleson, S. R., Poston, D. I., & McClure, P. R. (2017). NASA's Kilopower reactor development and the path to higher power missions. *IEEE Aerospace Conference Proceedings*.
15. Janek, J., & Zeier, W. G. (2016). A solid future for battery development. *Nature Energy*, 1(9), 16141.
16. NASA SBIR. (2024). Extreme temperature battery development for lunar applications. NASA Small Business Innovation Research Program.
17. NASA. (2022). Webb's orbit at Sun-Earth Lagrange Point 2 (L2). NASA Science Mission Directorate.
18. Ball, J. L., Mackie, S., van de Lindt, J. G., et al. (2025). Measurements of fusion yield on the Centrifugal Mirror Fusion Experiment. arXiv:2505.23047.
19. Dai, J., et al. (2022). Review of the development of the hydrogen maser technique and introduction to its space applications. *Frontiers in Physics*, 10, 995361.
20. Vessot, R. F. C., & Levine, M. W. (1979). A test of the equivalence principle using a space-borne clock. *General Relativity and Gravitation*, 10(3), 181-204.
21. Li, S., et al. (2018). Direct evidence of surface exposed water ice in the lunar polar regions. *Proceedings of the National Academy of Sciences*, 115(36), 8907-8912.
22. TAE Technologies. (2025). TAE Technologies delivers fusion breakthrough that dramatically reduces cost of a future power plant. Press Release, April 2025.
23. Interesting Engineering. (2025). China reports 824 Wh/kg solid-state EV battery, targets 1,000 Wh/kg. December 2025.